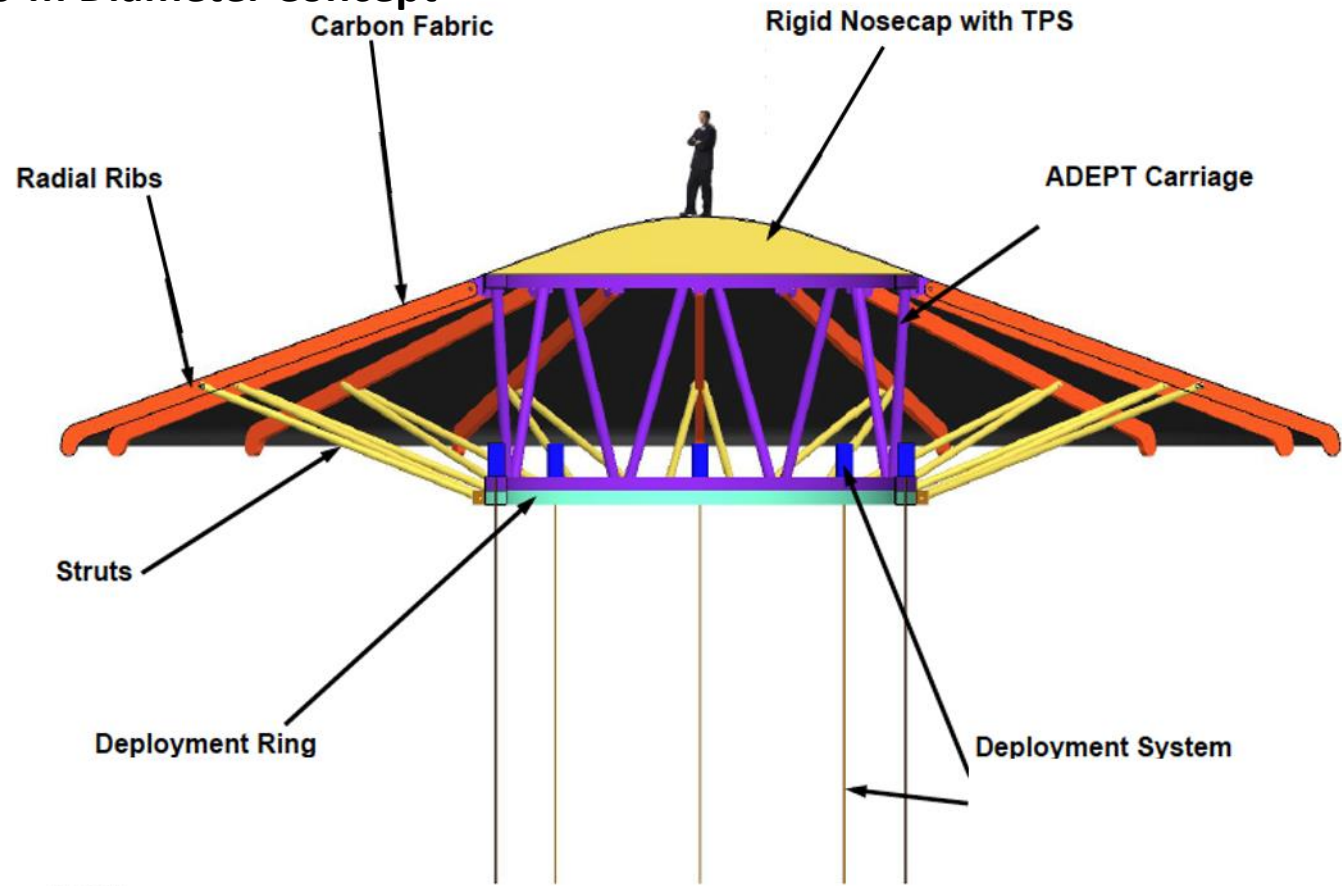


ADEPT 16-m Diameter Concept



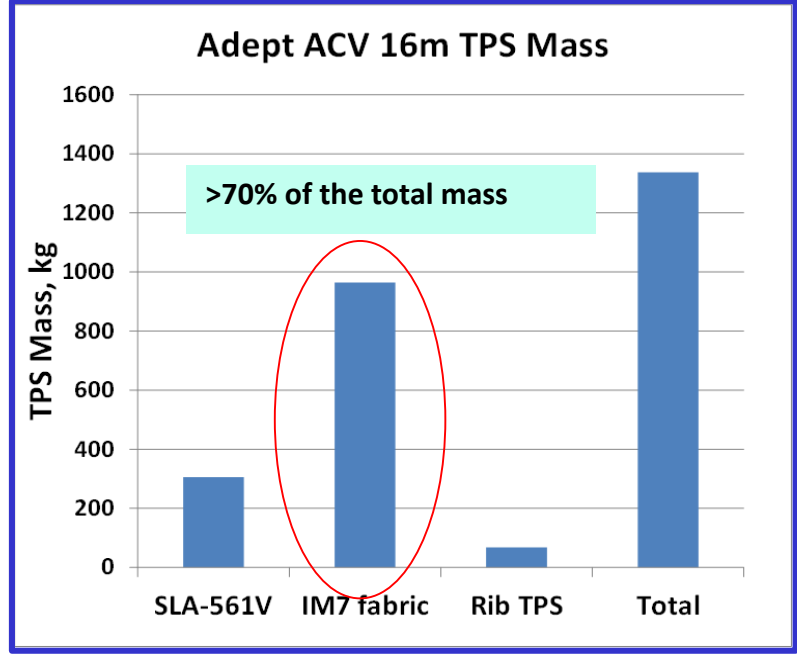
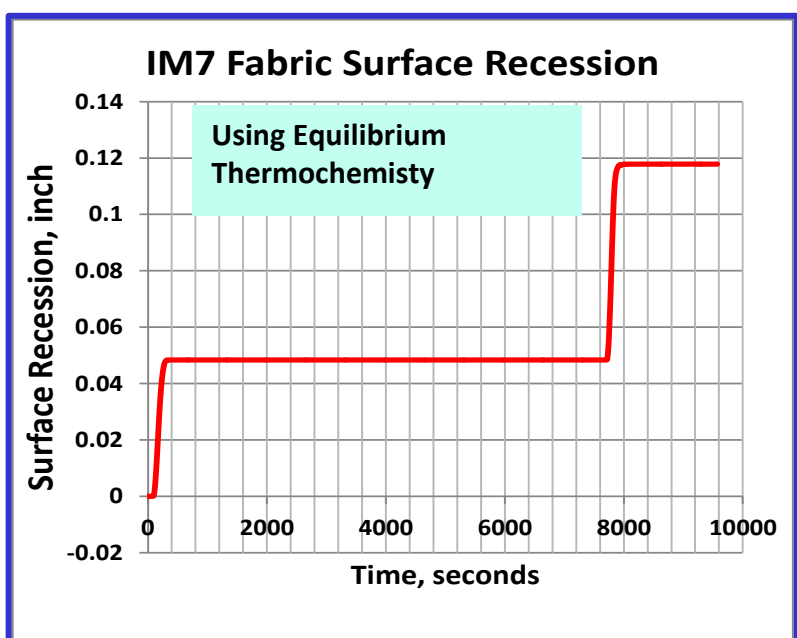
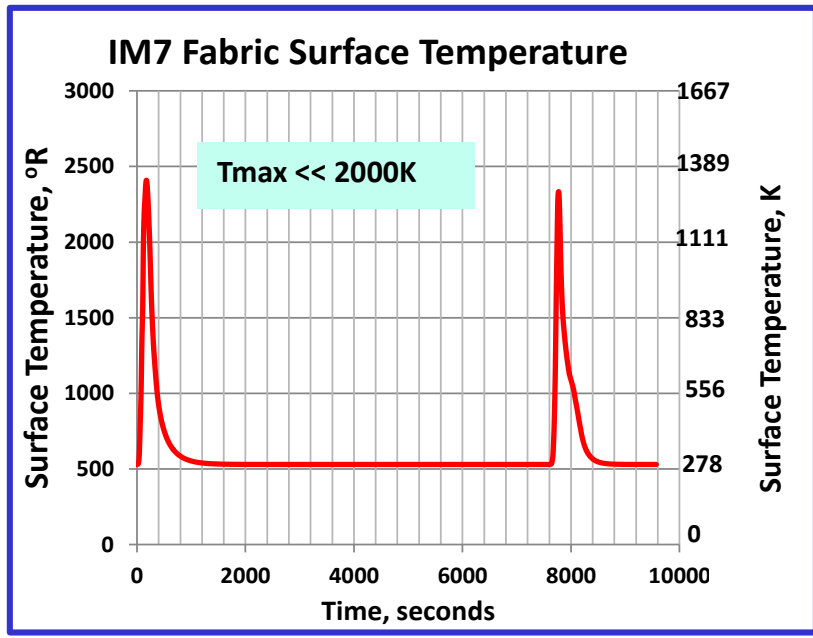
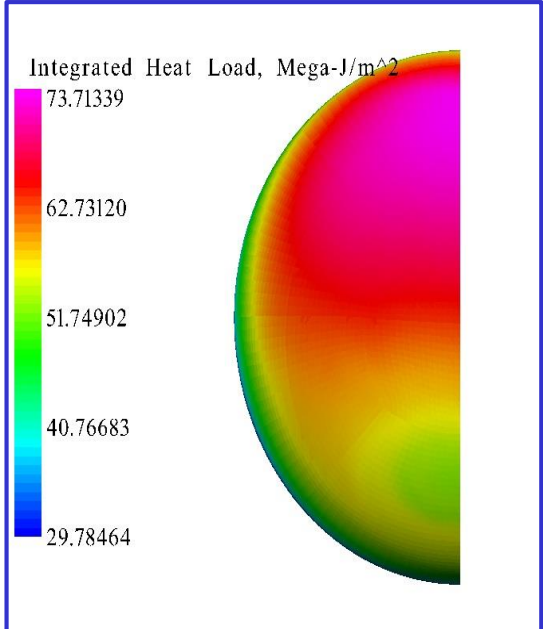
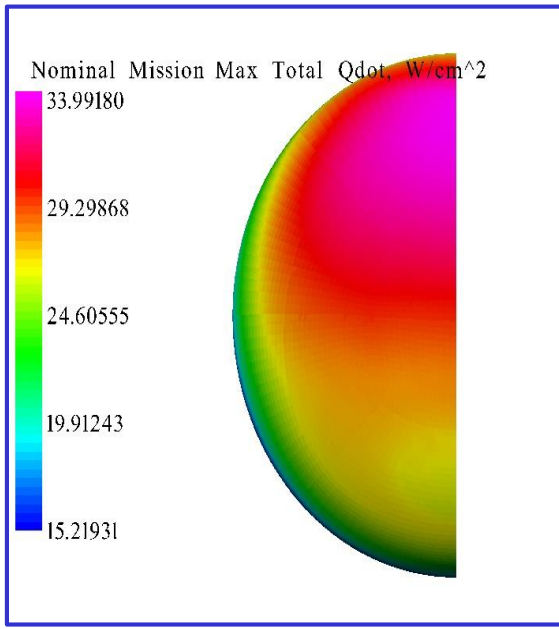
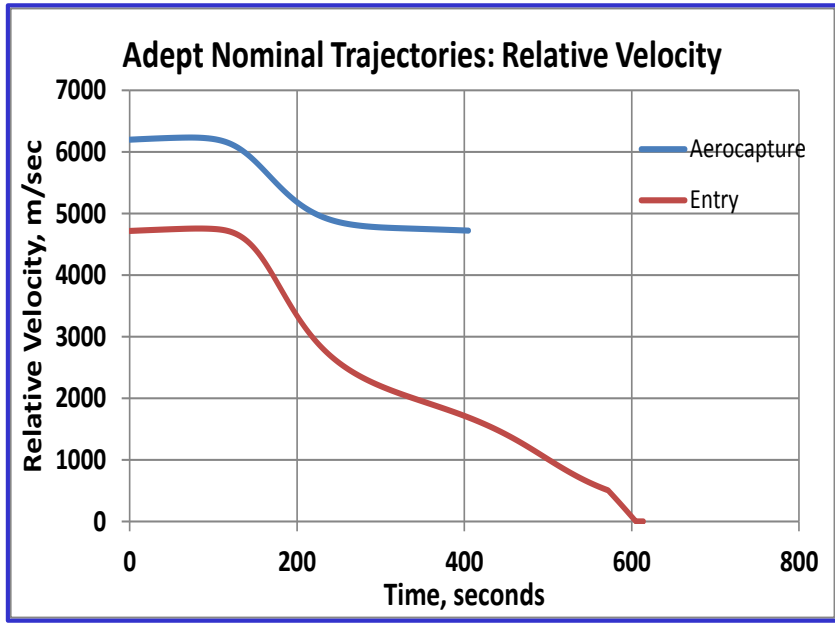
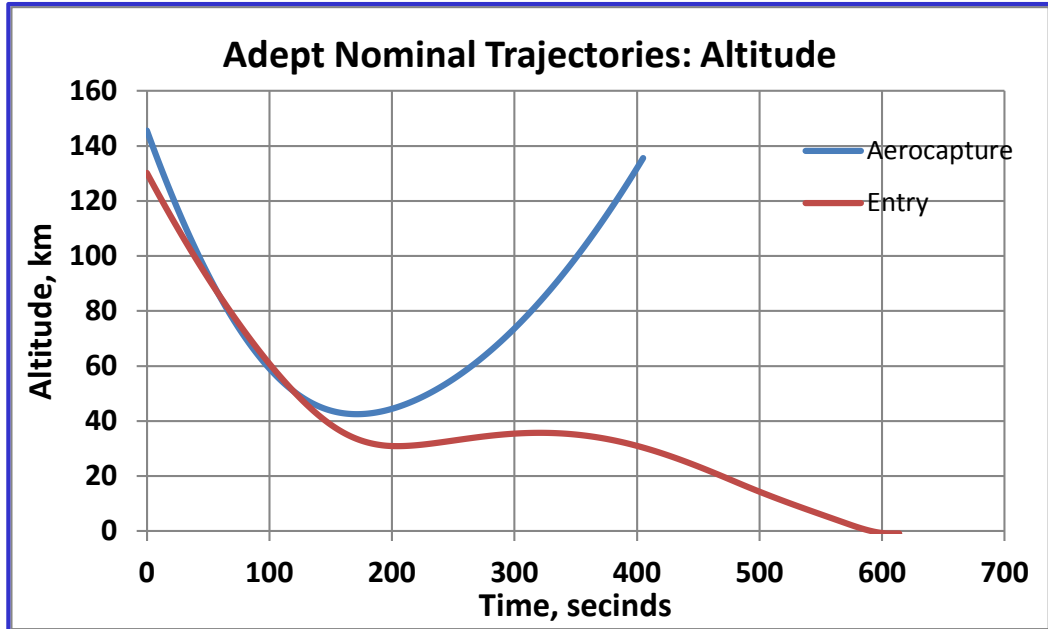
The Adaptive Deployable Entry and Placement Technology (ADEPT)

- A mechanically deployable decelerator is being considered as an entry, descent and landing (EDL) system to enable Human Mars class missions
- Ground rules for the Mars studies required aerocapture, orbit, and then entry
- Utilizes a 3-D woven carbon cloth fabric as both heatshield and primary structure
- Design guidelines required 6 layers remaining after all entry events

The Problem

- The peak heating predicted for the ADEPT carbon cloth is <35 W/cm² and resulting temperatures were predicted to be <1400K
- Predictions for carbon mass loss were performed using equilibrium thermochemistry, which is only accurate for T>2000K
- Carbon oxidation is kinetically controlled at T<2000K, and mass loss drops off considerably from equilibrium values
- Equilibrium predictions resulted in a 15-layer carbon cloth design, with the cloth representing ~70% of the TPS mass
- Design of the cloth thickness and *mass would be significantly reduced* if kinetics were considered, but development of the kinetic constants for Carbon in CO₂ would be costly and difficult to implement in the trade studies

1: Background



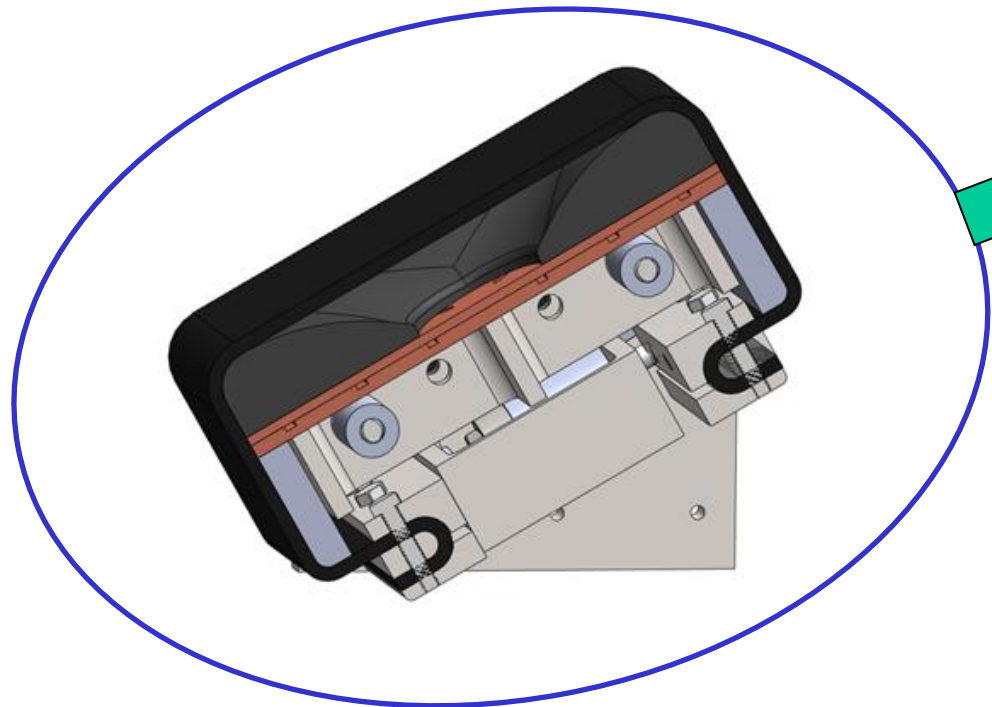
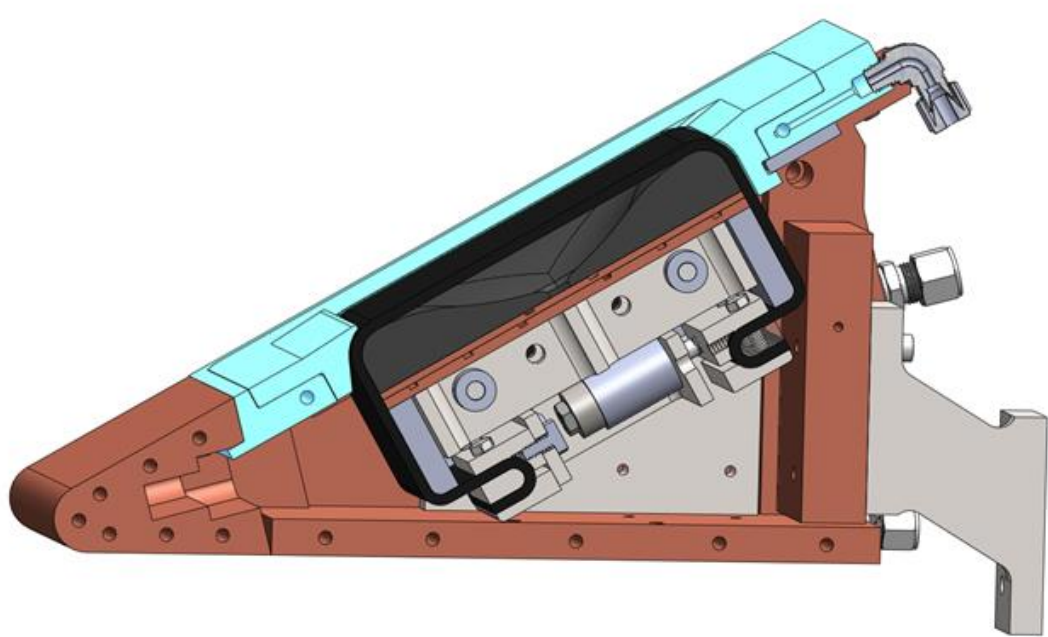
Figures from a report by Jeffrey Bowles (ARC), Steven Tobin (LaRC) and Stanley Bouslog (JSC)

2: Design Approach

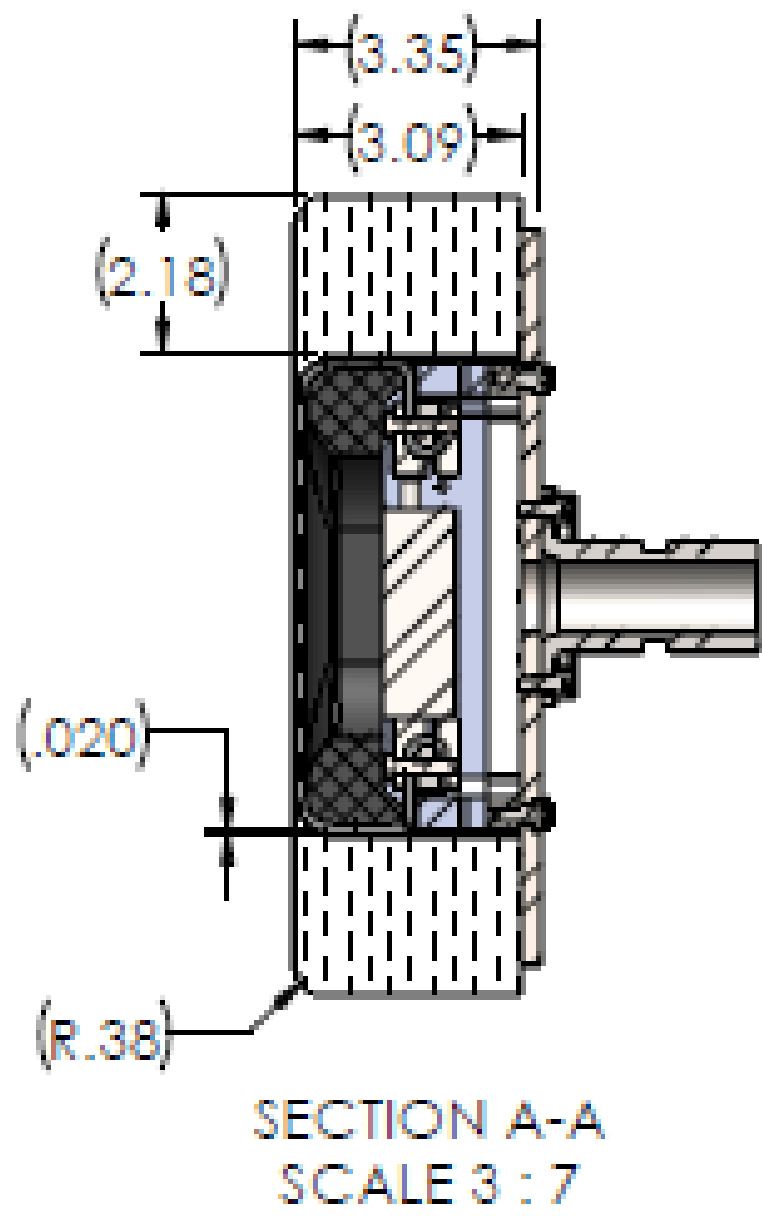
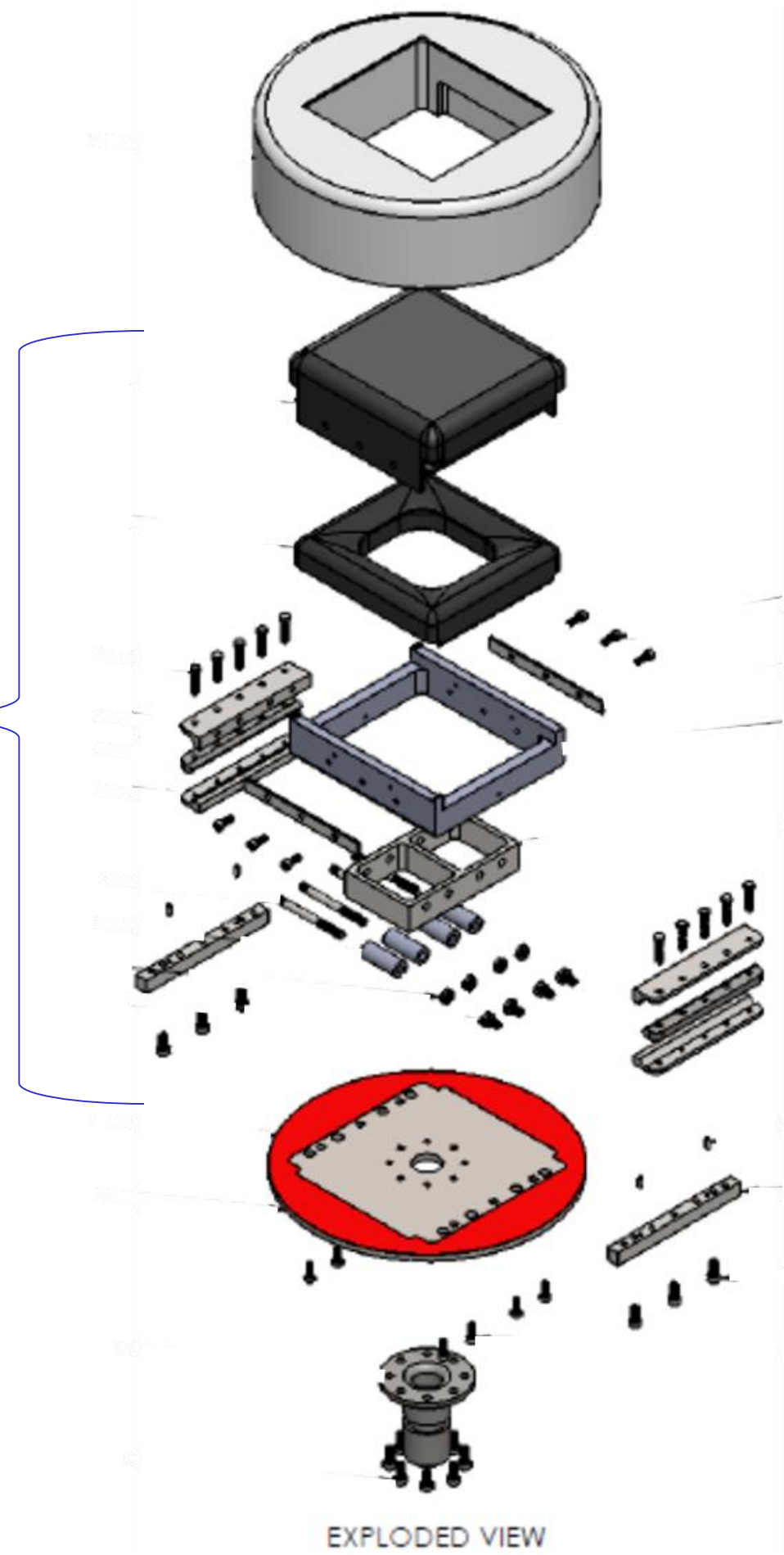
The Approach

- Repurpose an existing nearly square fabric tensioning design from earlier ADEPT testing (Bilaterally Loaded Ablation Model [BLAM]) that was used in a wedge holder to evaluate the response of the cloth in shear while under load
- Simplify the design by removing the load cell and therefore negating the need for a water cooled part
- Place an insulating collar around the tensioning section and a new back plate
- Build a prototype (in process)

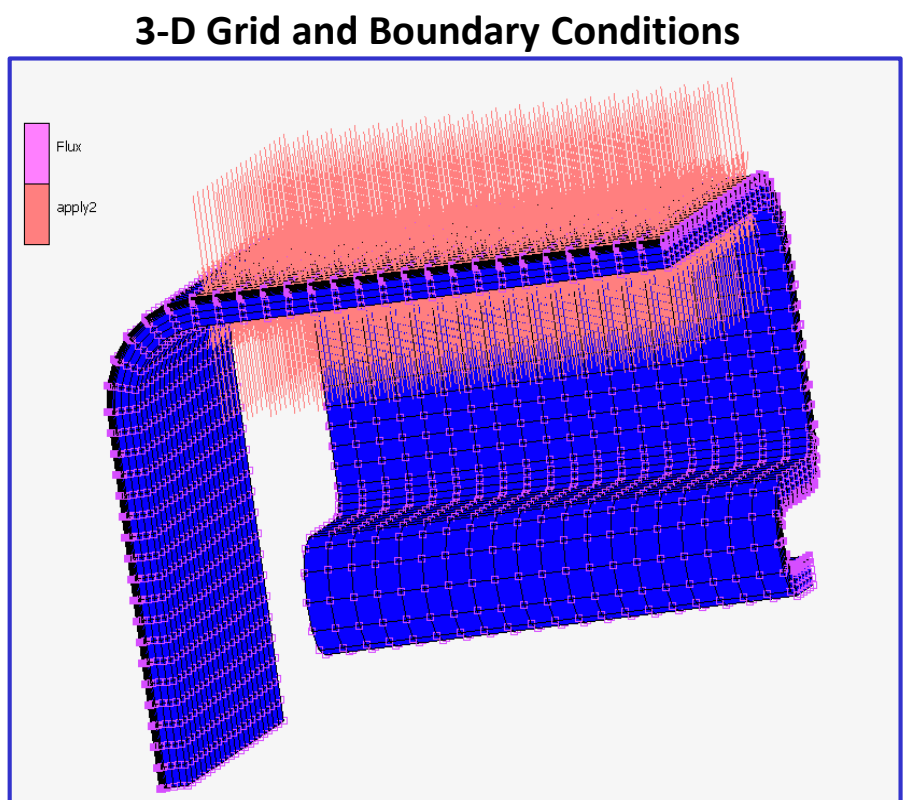
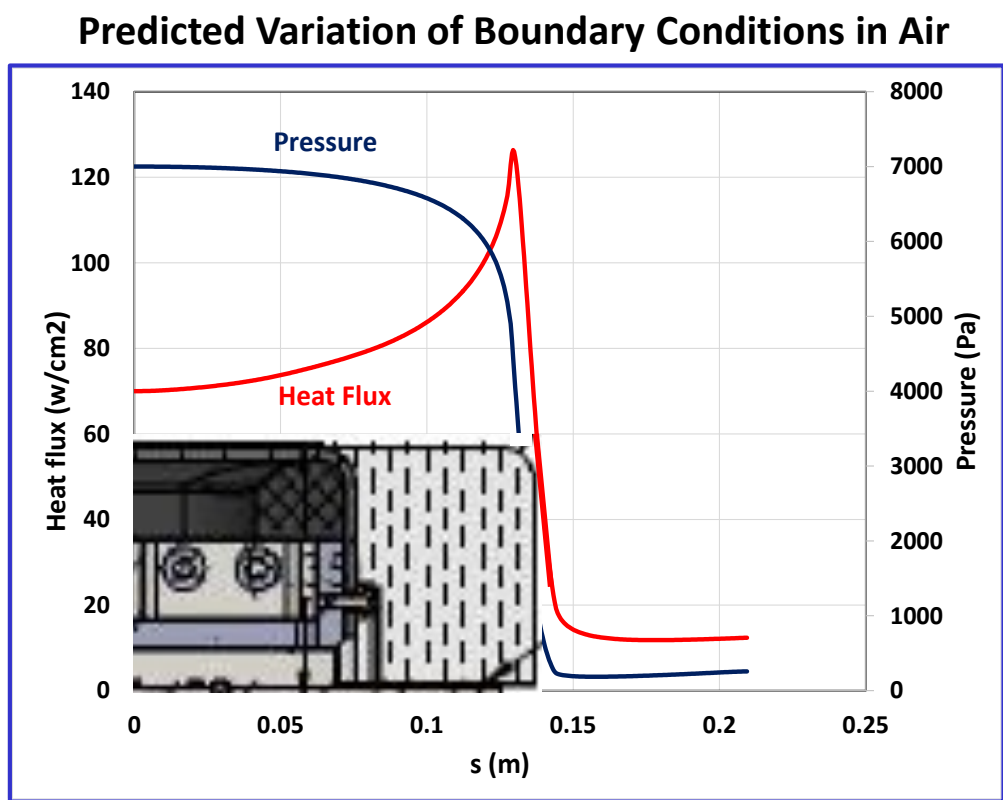
Original ADEPT BLAM design



Simplified, modified



3: Analysis



Analysis Approach

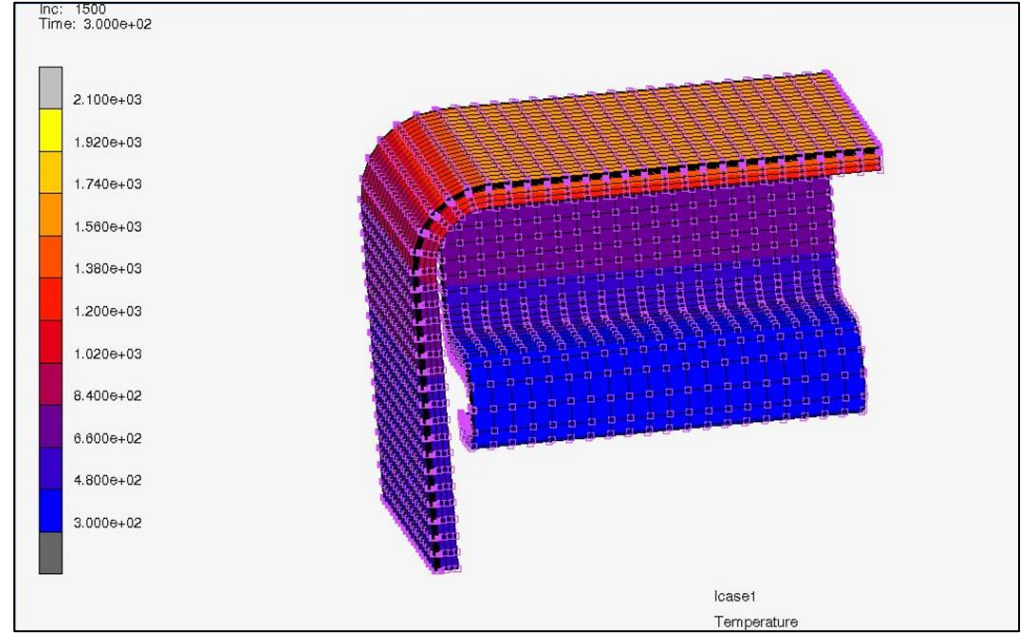
- CFD analysis of a typical AHF test condition on model with a target of ~70 W/cm² (2x predicted entry environments)
- Very conservative 3-D Finite Element model developed for the new carbon cloth design
 - 20,800 hex elements, 23814 nodes
 - 100 W/cm² and 50 W/cm² constant heatflux applied to top surface for 5 minutes, followed by 10 minute cooldown
 - Only top and bottom surface re-radiating to the environment, all other surfaces adiabatic
 - Transverse isotropic properties included for thermal modeling
 - Carbon cloth has much higher conductivity in-plane than through the thickness
- Future analysis work would include the Li2200 collar and the graphite frame beneath the cloth

The Results

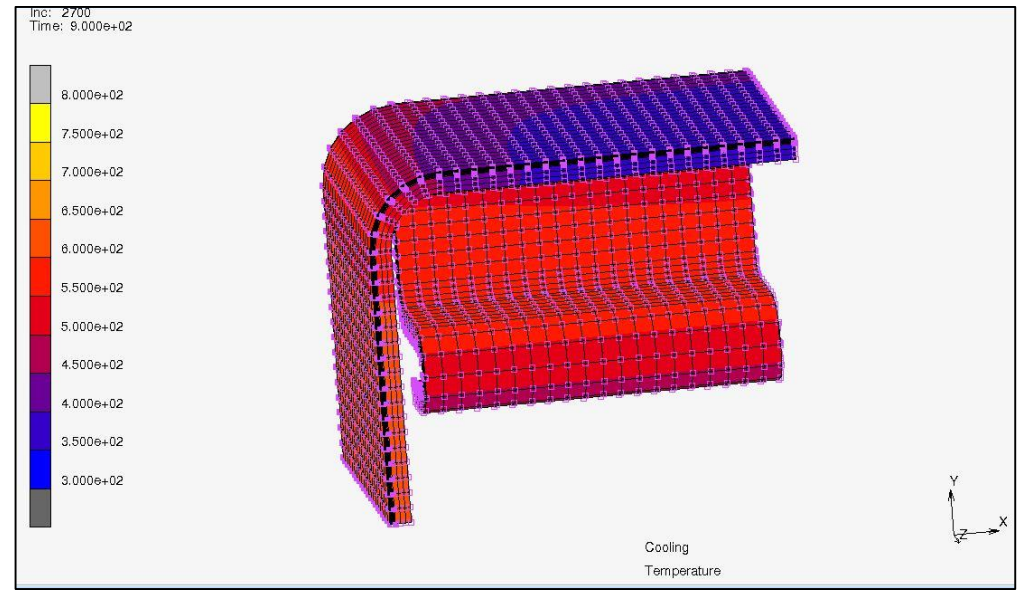
- Analysis shows that the collar material will survive heating due to the carbon cloth in proximity (T_{carbon}<<T_{melt} Li2200), as will all other materials in contact
- This design should work well in the AHF in flows with heatfluxes at or below 100 W/cm² with no loss of material integrity

50 W/cm², 5-min

Temperatures at end of ² exposure

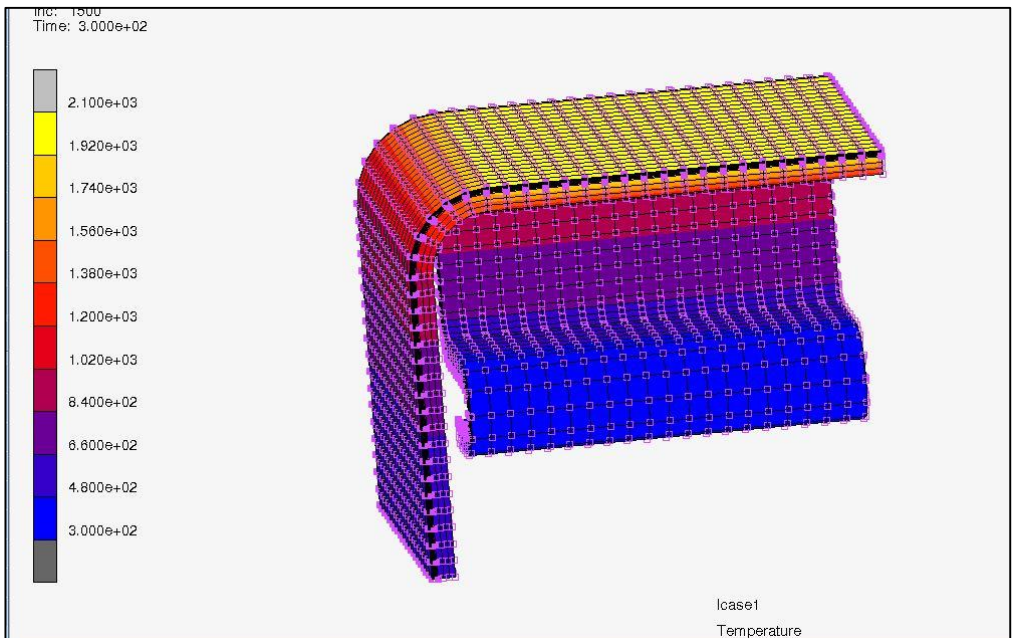


Temperatures at end 10-min cooldown

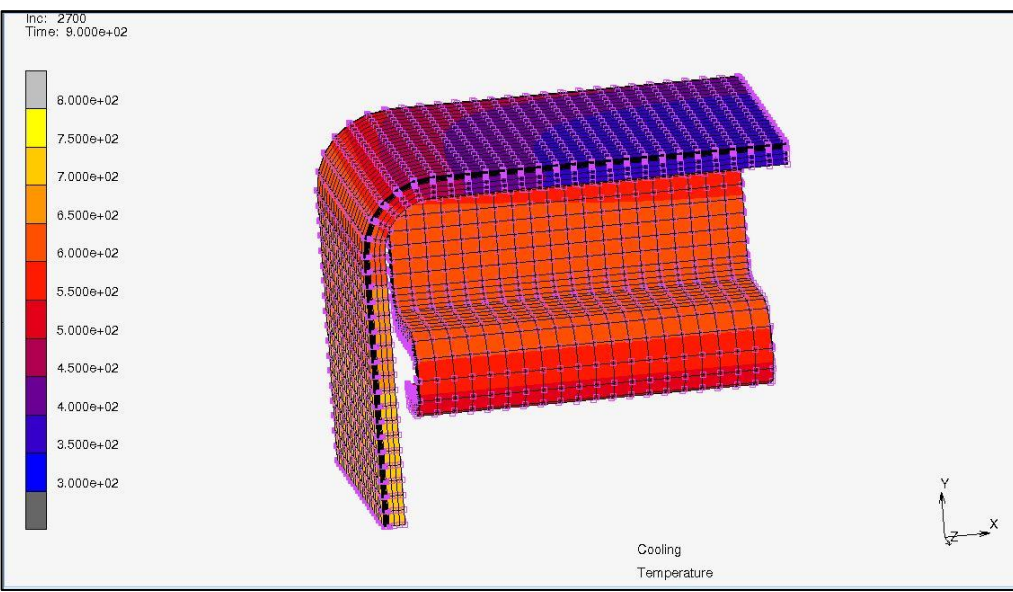


100 W/cm², 5-min

Temperatures at end of exposure



Temperatures at end 10-min cooldown



4: Summary

A new stagnation test article has been designed for developing an engineering model representing the mass loss of carbon cloth as a function of the partial pressure of monatomic oxygen for more reasonable predictions of carbon cloth thickness requirements in low heating environments

5: Acknowledgements

- This work was funded by NASA Ames FY17 Director's Discretionary Fund
- The Human Mars ADEPT analysis was funded by STMD GCD
- The original BLAM design was funded by STMD GCD